

AD-A172 784

STRESS RATES IN CONTINUUM MECHANICS AND COMPUTER CODES

1/1

(U) S-CUBED LA JOLLA CA D H BROWNELL 10 APR 86

555-R-86-7887 DNA-TR-86-121 DNA001-85-C-0024

UNCLASSIFIED

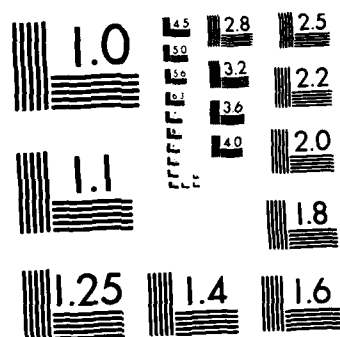
F/G 20/11

NL

[illegible]

100

100



MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

AD-A172 704

(12)
DNA-TR-86-121

STRESS RATES IN CONTINUUM MECHANICS AND COMPUTER CODES

**D. H. Brownell, Jr.
S-CUBED
A Division of Maxwell Laboratories, Inc.
P. O. Box 1620
La Jolla, CA 92038-1620**

**DTIC
ELECTE
OCT 14 1986
S D**

10 April 1986

Technical Report

CONTRACT No. DNA 001-85-C-0024

**Approved for public release;
distribution is unlimited.**

**THIS WORK WAS SPONSORED BY THE DEFENSE NUCLEAR AGENCY
UNDER RDT&E RMC CODE B3440854662 RS RA 00046 25904D.**

**Prepared for
Director
DEFENSE NUCLEAR AGENCY
Washington, DC 20305-1000**

FILE COPY

26 10 9 043

Destroy this report when it is no longer needed. Do not return to sender.

PLEASE NOTIFY THE DEFENSE NUCLEAR AGENCY,
ATTN: STTI, WASHINGTON, DC 20305-1000, IF YOUR
ADDRESS IS INCORRECT, IF YOU WISH IT DELETED
FROM THE DISTRIBUTION LIST, OR IF THE ADDRESSEE
IS NO LONGER EMPLOYED BY YOUR ORGANIZATION.



DISTRIBUTION LIST UPDATE

This mailer is provided to enable DNA to maintain current distribution lists for reports. We would appreciate your providing the requested information.

- ☐ Add the individual listed to your distribution list.
- ☐ Delete the cited organization/individual.
- ☐ Change of address.

NAME: _____

ORGANIZATION: _____

OLD ADDRESS

CURRENT ADDRESS

TELEPHONE NUMBER: () _____

SUBJECT AREA(s) OF INTEREST:

DNA OR OTHER GOVERNMENT CONTRACT NUMBER: _____

CERTIFICATION OF NEED-TO-KNOW BY GOVERNMENT SPONSOR (if other than DNA):

SPONSORING ORGANIZATION: _____

CONTRACTING OFFICER OR REPRESENTATIVE: _____

SIGNATURE: _____

Director
Defense Nuclear Agency
ATTN: STTI
Washington, DC 20305-1000

Director
Defense Nuclear Agency
ATTN: STTI
Washington, DC 20305-1000

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

AD-A122704

REPORT DOCUMENTATION PAGE

1a. REPORT SECURITY CLASSIFICATION UNCLASSIFIED			1b. RESTRICTIVE MARKINGS	
2a. SECURITY CLASSIFICATION AUTHORITY N/A since Unclassified			3. DISTRIBUTION/AVAILABILITY OF REPORT Approved for public release; distribution is unlimited.	
2b. DECLASSIFICATION/DOWNGRADING SCHEDULE N/A since Unclassified				
4. PERFORMING ORGANIZATION REPORT NUMBER(S) SSS-R-86-7887			5. MONITORING ORGANIZATION REPORT NUMBER(S) DNA-TR-86-121	
6a. NAME OF PERFORMING ORGANIZATION S-CUBED, A Division of Maxwell Laboratories, Inc.		6b. OFFICE SYMBOL (If applicable)	7a. NAME OF MONITORING ORGANIZATION Director Defense Nuclear Agency	
6c. ADDRESS (City, State, and ZIP Code) P. O. Box 1620 La Jolla, CA 92038-1620			7b. ADDRESS (City, State, and ZIP Code) Washington, DC 20305-1000	
8a. NAME OF FUNDING/SPONSORING ORGANIZATION		8b. OFFICE SYMBOL (If applicable)	9. PROCUREMENT INSTRUMENT IDENTIFICATION NUMBER DNA 001-85-C-0024	
8c. ADDRESS (City, State, and ZIP Code)			10. SOURCE OF FUNDING NUMBERS	
			PROGRAM ELEMENT NO. 62715H	PROJECT NO RS
			TASK NO RA	WORK UNIT ACCESSION NO. DH008681
11. TITLE (Include Security Classification) STRESS RATES IN CONTINUUM MECHANICS AND COMPUTER CODES				
12. PERSONAL AUTHOR(S) Brownell, D. H., Jr.				
13a. TYPE OF REPORT Technical Report		13b. TIME COVERED FROM 841101 TO 860401		14. DATE OF REPORT (Year, Month, Day) 860410
15. PAGE COUNT 24				
16. SUPPLEMENTARY NOTATION This work was sponsored by the Defense Nuclear Agency under RDT&E RMC Code B3440854662 RS RA 00046 25904D.				
17. COSATI CODES			18. SUBJECT TERMS (Continue on reverse if necessary and identify by block number)	
FIELD	GROUP	SUB-GROUP		
20	11		Stress Rates Constitutive Relations Plasticity	
9	2		Objectivity Frame Indifference	
19. ABSTRACT (Continue on reverse if necessary and identify by block number) The concept of an objective (frame-indifferent) stress rate is introduced. The behavior of several objective stress rates in a material undergoing simple extension is discussed. The response of a hypoelastic material in simple shear, in both the elastic and elastic/plastic regimes, is evaluated, and it is concluded that for materials with low yield strength, such as metals and geologic materials, use of the Jaumann stress rate is justified. Key words:				
20. DISTRIBUTION/AVAILABILITY OF ABSTRACT <input type="checkbox"/> UNCLASSIFIED/UNLIMITED <input checked="" type="checkbox"/> SAME AS RPT <input type="checkbox"/> OTIC USERS			21. ABSTRACT SECURITY CLASSIFICATION UNCLASSIFIED	
22a. NAME OF RESPONSIBLE INDIVIDUAL Betty L. Fox			22b. TELEPHONE (Include Area Code) 202 325-7042	22c. OFFICE SYMBOL DNA/STI

DD FORM 1473, 84 MAR

83 APR edition may be used until exhausted.
All other editions are obsolete.SECURITY CLASSIFICATION OF THIS PAGE
UNCLASSIFIED

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE



UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

TABLE OF CONTENTS

Section	Page
1 Stress Rates in Continuum Mechanics and Computer Codes.....	1
2 List of References.....	14

SECTION 1

The necessity of paying careful attention to stress rates in the theory of mechanics of a continuous medium is illustrated by a simple example: Suppose a body undergoes rigid body motion (rotation and translation) without stretching, i.e.,

$$D_{ij} = \frac{1}{2} \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right) = 0 \quad (1)$$

where \underline{x} = position, \underline{v} = velocity and \underline{D} is the rate-of-deformation tensor. Then, at every point in the body, in general, the time derivative of the stress tensor is constant only in a frame of reference attached to the body, i.e., a rotating frame. In other words,

$$\dot{a} = \frac{Da}{Dt} \neq 0 \quad (2)$$

In this case of rigid body motion, the quantity

$$\underline{\underline{\sigma}}_1 = \underline{\underline{\dot{\sigma}}} - \underline{\underline{W}} \underline{\underline{\sigma}} + \underline{\underline{\sigma}} \underline{\underline{W}} \quad (3)$$

actually is = 0; here \underline{W} is the spin tensor:

$$w_{ij} = \frac{1}{2} \left(\frac{\partial v_i}{\partial x_j} - \frac{\partial v_j}{\partial x_i} \right) \quad (4)$$

The quantity $\underline{\dot{\sigma}}_J$ is called the Jaumann stress rate, and is one of an infinite family of stress rates which satisfy the principle of objectivity, or frame indifference. In addition to these objective stress rates, other objective tensor quantities are \underline{D} , scalars, and

the so-called Rivlin-Ericksen tensors; examples of non-frame-indifferent tensor quantities are \underline{W} , \underline{v} and $\underline{\dot{\sigma}}$. For a discussion of frame-indifference, see [1]. The objective stress rate typically appears in a constitutive relation of the form

$$\dot{\sigma}_{ij} = C_{ijkl} [D_{kl} - D_{kl}^P] \quad (5)$$

where \underline{C} is the elastic constitutive tensor and \underline{D}^P the plastic strain rate.

Stress rates which are closely related to the Jaumann rate and which have appeared in the literature are the Cotter-Rivlin rate:

$$\dot{\sigma}_{CR} \equiv \dot{\sigma}_J + \underline{D} \underline{\sigma} + \underline{\sigma} \underline{D} \quad (6)$$

the Oldroyd rate:

$$\dot{\sigma}_O \equiv \dot{\sigma}_J - (\underline{D} \underline{\sigma} + \underline{\sigma} \underline{D}) \quad (7)$$

and the Truesdell rate:

$$\dot{\sigma}_T \equiv \dot{\sigma}_O + \underline{\sigma} \text{tr} \underline{D} \quad (8)$$

These rates are defined in terms of tensor quantities evaluated in the current material configuration (one point tensors), and are easy to incorporate in either an Eulerian or Lagrangian computer code. An objective stress rate which is gaining popularity in Lagrangian codes is the Green-Naghdi rate:

$$\dot{\sigma}_{GN} = \dot{\sigma} - \underline{\Omega} \underline{\sigma} + \underline{\sigma} \underline{\Omega} \quad (9)$$

where

$$\underline{\Omega} \equiv \dot{\underline{R}} \underline{R}^T \quad (10)$$

and \underline{R} is the proper-orthogonal rotation tensor in the polar decomposition of the deformation gradient:

$$\underline{\underline{F}} = \underline{\underline{R}} \underline{\underline{U}} \quad (11)$$

where

$$F_{ij} = \partial x_i / \partial X_j \quad (12)$$

Here $\underline{\underline{X}}$ is the position of the particle, currently at $\underline{\underline{x}}$, in the undeformed body; $\underline{\underline{F}}$ is a so-called two point tensor. It is easily seen that the evaluation of $\dot{\underline{\underline{a}}}_{GN}$ requires "keeping books" on the original position of each material point in the continuum, and while feasible in Lagrangian codes would be considerably more complicated in an Eulerian context.

In [2] it is shown that of the Jaumann, Oldroyd, Cotter-Rivlin and Truesdell rates, only the Jaumann rate gives a linear stress-strain response for a hypoelastic body of grade zero undergoing simple extension, i.e., for

$$\underline{\underline{D}} = \begin{bmatrix} \dot{\eta} & 0 & 0 \\ 0 & \dot{\eta} & 0 \\ 0 & 0 & \dot{\epsilon} \end{bmatrix} \quad (13)$$

where η and ϵ are functions of time (see Figure 1). In addition, it is shown in [3] that of the Jaumann, Oldroyd, Cotter-Rivlin and Truesdell stress rates, only the Jaumann rate has the property that when it vanishes the stress invariants are stationary.

However, it is shown in [4] that the Jaumann rate suffers a serious deficiency: consider a linear elastic medium undergoing pure rectilinear shear (plane strain), with

$$\underline{\underline{D}} = \begin{bmatrix} 0 & a/2 & 0 \\ a/2 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad (14)$$

$$\underline{\underline{W}} = \begin{bmatrix} 0 & a/2 & 0 \\ -a/2 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad (15)$$

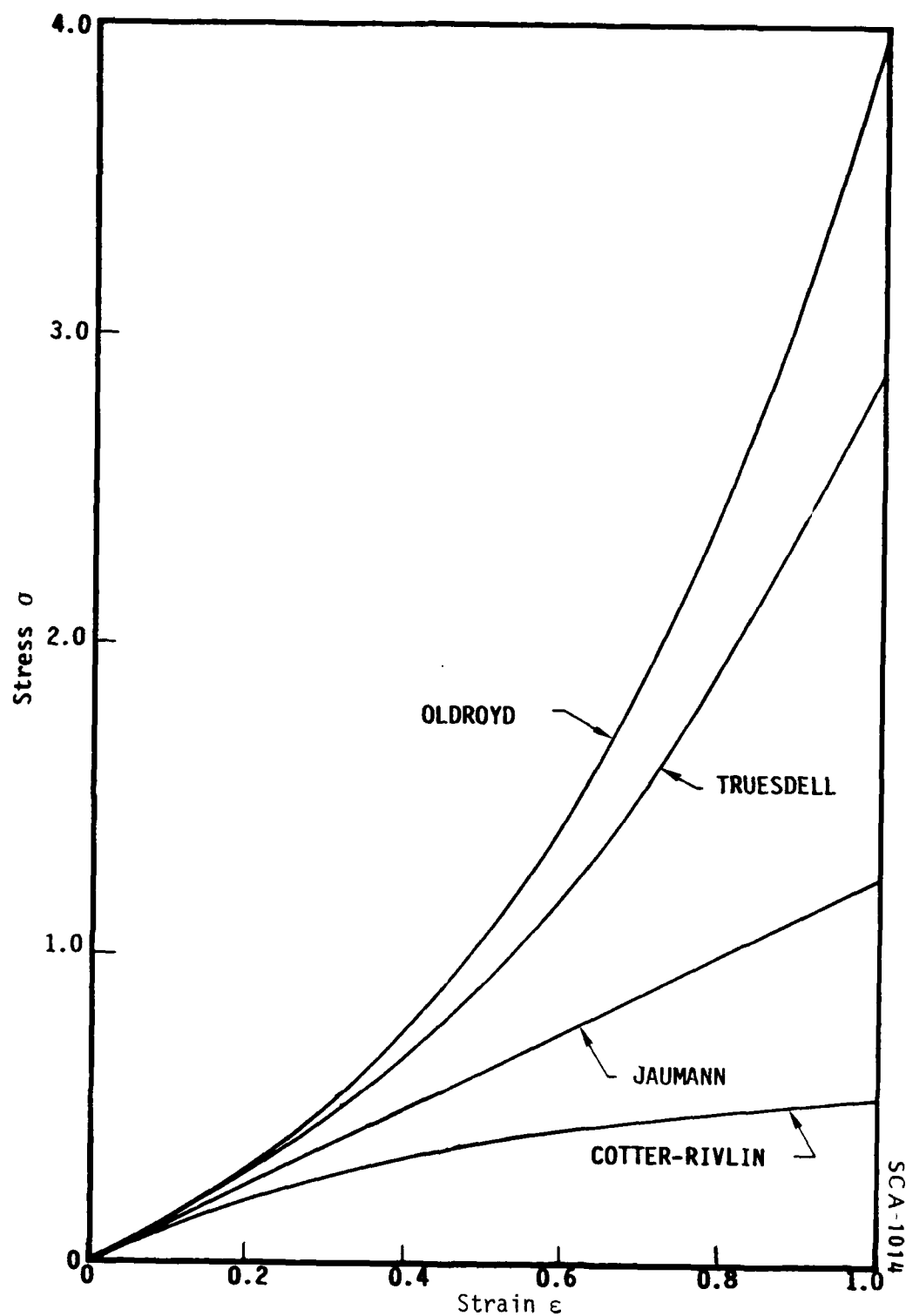


Figure 1. The constitutive relations for a hypoelastic body of grade zero undergoing simple extension for Poisson ratio = 1.4).

and obeying the constitutive relation

$$\dot{\underline{\underline{\sigma}}}_J = \lambda \underline{\underline{I}} \text{tr } \underline{\underline{D}} + 2\mu \underline{\underline{D}} \quad (16)$$

where λ and μ are the Lamé constants and $\underline{\underline{I}}$ is the identity tensor, with stress and strain equal zero initially, and $a = \text{constant}$. The governing equations are then

$$\dot{\sigma}_{11} - a \sigma_{12} = 0 \quad (17a)$$

$$\dot{\sigma}_{12} - \frac{1}{2} a (\sigma_{22} - \sigma_{11}) = \mu a \quad (17b)$$

$$\dot{\sigma}_{22} + a \sigma_{12} = 0 \quad (17c)$$

$$\dot{\sigma}_{33} = 0 \quad (17d)$$

which have the solution (for the above initial conditions)

$$\sigma_{12} = \mu \sin at \quad (18a)$$

$$\sigma_{11} = \mu(1 - \cos at) \quad (18b)$$

$$\sigma_{22} = -\sigma_{11} \quad (18c)$$

$$\sigma_{33} = 0 \quad (18d)$$

The sinusoidal variation of stress is clearly physically unreasonable. On the other hand, it is shown in [4] that the use of $\dot{\underline{\underline{\sigma}}}_{GN}$ instead of $\dot{\underline{\underline{\sigma}}}_J$ in (16) gives a monotonic behavior of stress for this problem (Figure 2). This fact has led to the implementation of the Green-Naghdi rate in Lagrangian codes such as NIKE2D, DYNA2D and HONDO.

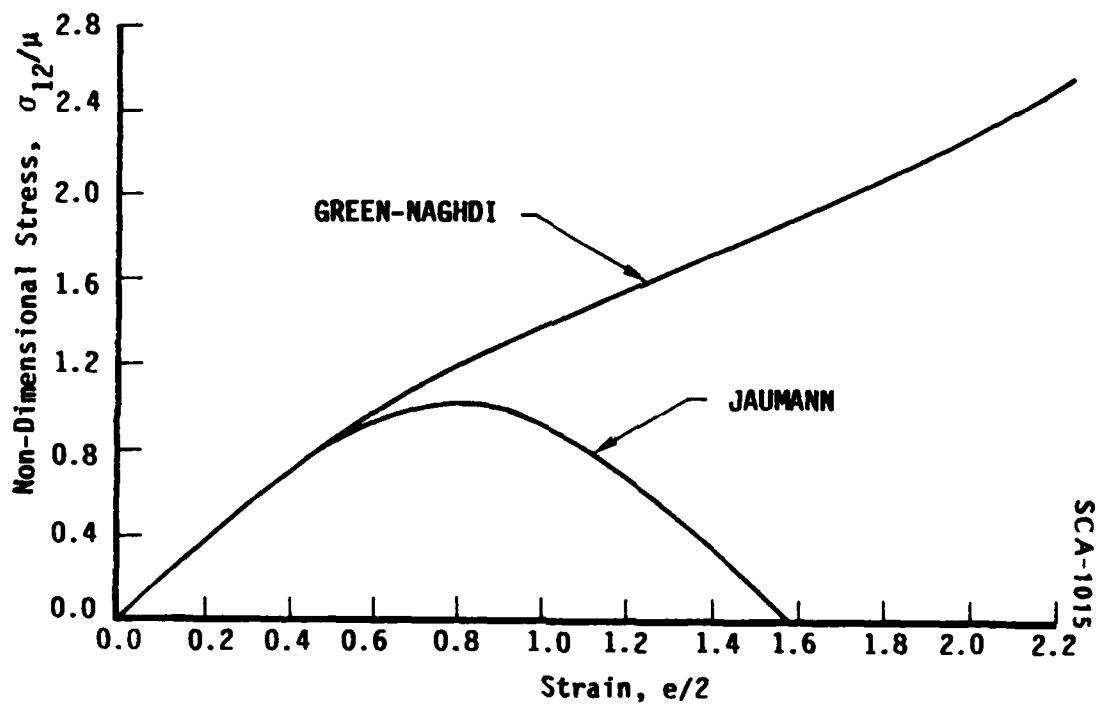


Figure 2. A comparison between the shear stress on a hypo-elastic material in a simple, rectilinear shear using the Green-Naghdi and the Jaumann stress rates.

Consider the above model problem of rectilinear shear with a stress rate of the form

$$\dot{\underline{\underline{\sigma}}} = \dot{\underline{\underline{\sigma}}} - \underline{\underline{W}} \underline{\underline{\sigma}} + \underline{\underline{\sigma}} \underline{\underline{W}} + \alpha (\underline{\underline{\sigma}} \underline{\underline{D}} + \underline{\underline{D}} \underline{\underline{\sigma}}) + \beta \underline{\underline{\sigma}} \text{tr} \underline{\underline{D}} \quad (19)$$

where α and β are constants. Note that the Cotter-Rivlin, Oldroyd and Truesdell rates are included in this form. For this particular problem the governing equations are

$$\dot{\sigma}_{11} + (\alpha - 1) \alpha \sigma_{12} = 0 \quad (20a)$$

$$\dot{\sigma}_{12} + \frac{1}{2} [(\alpha + 1) \alpha \sigma_{11} + (\alpha - 1) \alpha \sigma_{22}] = \mu \alpha \quad (20b)$$

$$\dot{\sigma}_{22} + (\alpha + 1) \alpha \sigma_{12} = 0 \quad (20c)$$

$$\dot{\sigma}_{33} = 0 \quad (20d)$$

Thus

$$\ddot{\sigma}_{12} = (\alpha - 1)(\alpha + 1) \alpha^2 \sigma_{12} \quad (21)$$

For the case $(\alpha - 1)(\alpha + 1) = 0$ the solution is

$$\sigma_{12} = \mu \alpha t \quad (22a)$$

$$\sigma_{11} = 0 \quad (\alpha = 1) \quad (22b)$$

$$= \alpha^2 \mu t^2 \quad (\alpha = -1)$$

$$\sigma_{22} = -\alpha^2 \mu t^2 \quad (\alpha = 1) \quad (22c)$$

$$= 0 \quad (\alpha = -1)$$

For the case $(a - 1)(a + 1) > 0$ the solution is

$$\sigma_{12} = \left\{ \mu / \left[(a - 1)(a + 1) \right]^{1/2} \right\} \sinh \left\{ \left[(a - 1)(a + 1) \right]^{1/2} at \right\} \quad (23a)$$

$$\sigma_{11} = \left[\mu / (a + 1) \right] \left[1 - \cosh \left\{ \left[(a - 1)(a + 1) \right]^{1/2} at \right\} \right] \quad (23b)$$

$$\sigma_{22} = \left[\mu / (a - 1) \right] \left[1 - \cosh \left\{ \left[(a - 1)(a + 1) \right]^{1/2} at \right\} \right] \quad (23c)$$

and for the case $(a - 1)(a + 1) < 0$ the solution is

$$\sigma_{12} = \left\{ \mu / \left[- (a - 1)(a + 1) \right]^{1/2} \right\} \sin \left\{ \left[- (a - 1)(a + 1) \right]^{1/2} at \right\} \quad (24a)$$

$$\sigma_{11} = \left[\mu / (a + 1) \right] \left[1 - \cos \left\{ \left[- (a - 1)(a + 1) \right]^{1/2} at \right\} \right] \quad (24b)$$

$$\sigma_{22} = \left[\mu / (a - 1) \right] \left[1 - \cos \left\{ \left[- (a - 1)(a + 1) \right]^{1/2} at \right\} \right] \quad (24c)$$

In all cases, $\sigma_{33} = 0$. All of the above solutions (22-24) have terms which are exponentially growing, sinusoidal (non-monotonic) or of order t^2 , all of which seem physically unreasonable.

However, the departure from "reasonable" behavior in those solutions occurs only after the deformation is very large, i.e., after the shear strain becomes of order unity. For metals and geologic materials, plastic failure occurs long before this point, so it is useful to investigate this problem in an elastic-plastic context. Suppose the governing equations are the above in the elastic regime, with the Jaumann stress rate, and, in the plastic regime, the associated flow rule

$$\underline{\underline{D}}^P = \Lambda \underline{\underline{S}} \quad (25)$$

where $\underline{\underline{D}}^P$ is the plastic strain rate:

$$\underline{\underline{D}} = \underline{\underline{D}}^e + \underline{\underline{D}}^P \quad (26)$$

where $\underline{\underline{D}}^e$ is the elastic strain rate, and

$$\underline{\underline{\sigma}}_J = \lambda \underline{\underline{I}} \text{tr } \underline{\underline{D}}^e + 2\mu \underline{\underline{D}}^e \quad (27)$$

with yield function

$$f = \frac{1}{2} S_{ij} S_{ij} - \frac{1}{3} Y^2 \quad (28)$$

where Y is the (constant) stress difference in compression for a von Mises material and S is the stress deviator:

$$S_{ij} = \sigma_{ij} - \frac{1}{3} \delta_{ij} \sigma_{kk} \quad (29)$$

This is the von Mises failure surface and the Prandtl-Reuss flow rule. In the (initial) elastic solution, $J_2' = 1/2 S_{ij} S_{ij}$ reaches a maximum at dimensionless time $t = \pi$, and we suppose plastic failure occurs before this point. Subsequently, as long as the material is on the yield surface, the governing equations are

$$\dot{\sigma}_{11} - a \sigma_{12} = -2\mu \wedge S_{11} \quad (30a)$$

$$\dot{\sigma}_{12} + \frac{1}{2} a (\sigma_{11} - \sigma_{22}) = \mu a - 2\mu \wedge S_{12} \quad (30b)$$

$$\dot{\sigma}_{22} + a \sigma_{12} = -2\mu \wedge S_{22} \quad (30c)$$

$$\dot{\sigma}_{33} = 2\mu \wedge (S_{11} + S_{22}) \quad (30d)$$

Adding (30a), (30c) and (30d), we obtain

$$\begin{aligned} \dot{\sigma}_{11} + \dot{\sigma}_{22} + \dot{\sigma}_{33} &= \text{tr } \dot{\sigma} \\ &= 0 \end{aligned} \quad (31)$$

and since at the end of the elastic solution, $\text{tr } \underline{g} = 0$, we have also for the plastic solution $\text{tr } \underline{g} = 0$, so that $\underline{g} = \underline{S}$.

Now

$$\begin{aligned} J_2' &= S_{11}^2 + S_{22}^2 + S_{12}^2 + S_{11} S_{22} \\ &= \frac{1}{3} \gamma^2 \end{aligned} \quad (32)$$

so that

$$2S_{11}\dot{S}_{11} + 2S_{22}\dot{S}_{22} + 2S_{12}\dot{S}_{12} + S_{11}\dot{S}_{22} + S_{22}\dot{S}_{11} = 0 \quad (33)$$

which gives

$$-4\mu \Lambda \left[S_{11}^2 + S_{22}^2 + S_{12}^2 + S_{11} S_{22} \right] + 2\mu a S_{12} = 0 \quad (34)$$

so that

$$\Lambda = 3a S_{12} / 2\gamma^2 \quad (35)$$

Also,

$$\begin{aligned} \dot{S}_{11} + \dot{S}_{22} &= -2\mu \Lambda (S_{11} + S_{22}) \\ &= \frac{3a\mu}{\gamma^2} (S_{11} + S_{22}) S_{12} \end{aligned} \quad (36)$$

Since $S_{11} = -S_{22}$ at the end of the elastic solution, (36) is satisfied thereafter if $S_{11} = -S_{22}$, so that the governing equations in the plastic flow regime become

$$\dot{S}_{11} - a S_{12} = -\frac{3a\mu}{\gamma^2} S_{11} S_{12} \quad (37a)$$

$$\dot{S}_{12} + a S_{11} = \mu a - \frac{3a\mu}{\gamma^2} S_{12}^2 \quad (37b)$$

$$S_{22} = -S_{11} \quad (37c)$$

$$S_{33} = 0 \quad (37d)$$

If a steady state exists for the equations (37), it is

$$S_{11} = \frac{\gamma^2}{3\mu} \quad (38a)$$

$$S_{12} = \left[\frac{Y^2}{3} \left(1 - \frac{Y^2}{3\mu^2} \right) \right]^{1/2} \quad (38b)$$

so that for a steady state to exist, we must have

$$Y \leq 3^{1/2} \mu \quad (39)$$

The character of the solution is as follows: if: $Y \geq 12^{1/2} \mu$ the elastic solution (18) persists forever. If $Y < 12^{1/2} \mu$, the elastic solution holds until dimensionless time

$$at_0 = \cos^{-1} \left[1 - \frac{Y^2}{6\mu^2} \right] \quad (40)$$

at which time

$$S_{11}(t_0) = \frac{Y^2}{6\mu} \quad (41a)$$

$$S_{12}(t_0) = Y \left[\left(1 - Y^2/12\mu^2 \right) / 3 \right]^{1/2} \quad (41b)$$

If $Y \leq 3^{1/2} \mu$, the solution asymptotes to (38); otherwise the solution oscillates between the elastic and plastic flow regimes. This has been verified by numerical solution of (37). Figure 3 shows the solution for the case $Y = \mu = 100$ kb; here $S_{12}(\infty)/S_{12}(t_0) = 0.85$. In the case $Y = \mu/10$ (see Figure 4), $S_{12}(\infty)/S_{12}(t_0) = 0.9987$. In all cases where a steady state exists, $S_{11}(\infty) = 2S_{11}(t_0)$.

These results indicate that for real materials with $Y \ll \mu$, the Jaumann stress rate should be adequate.

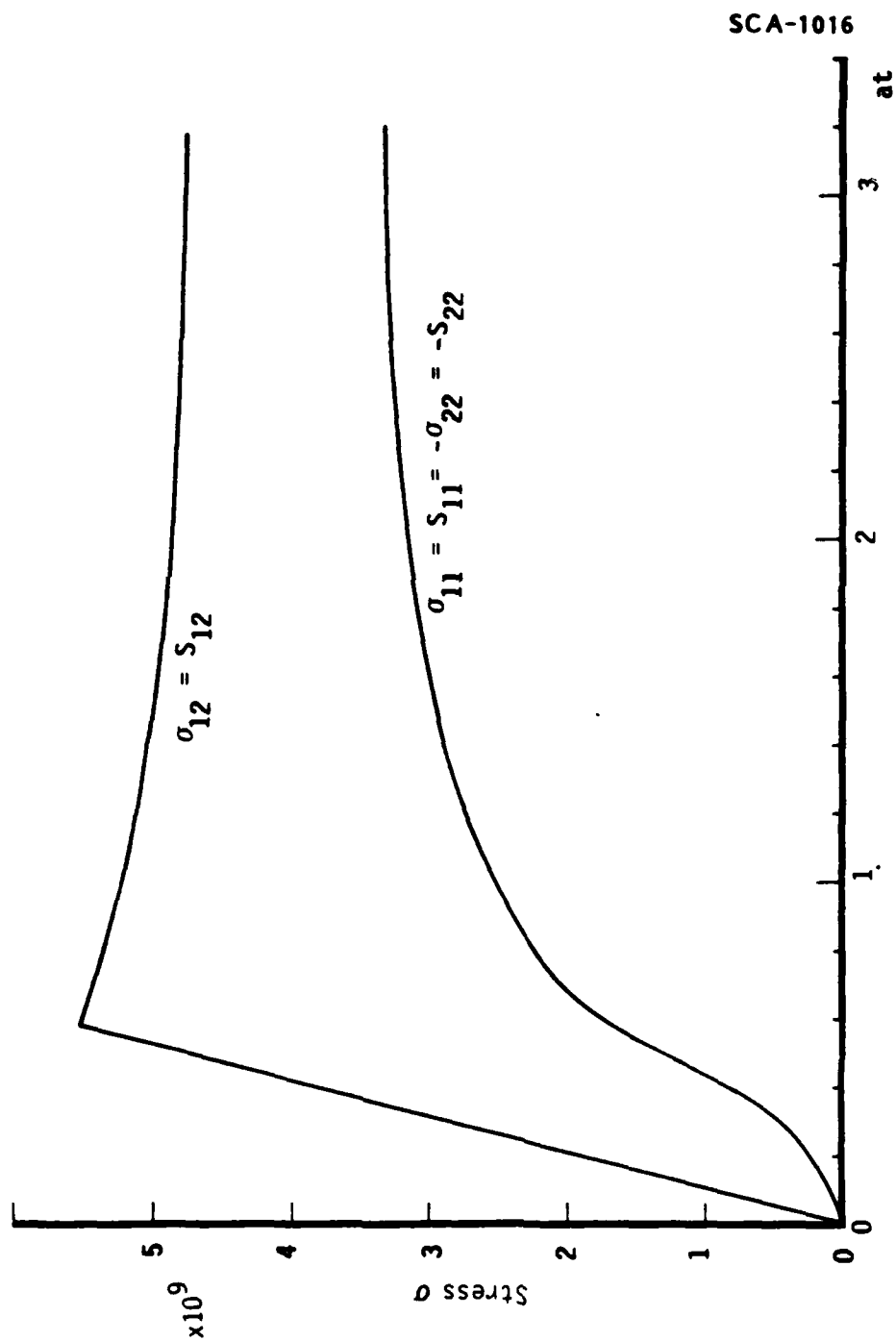


Figure 3. Elastic/plastic solution for $\gamma = \mu$: Stress vs. dimensionless time.

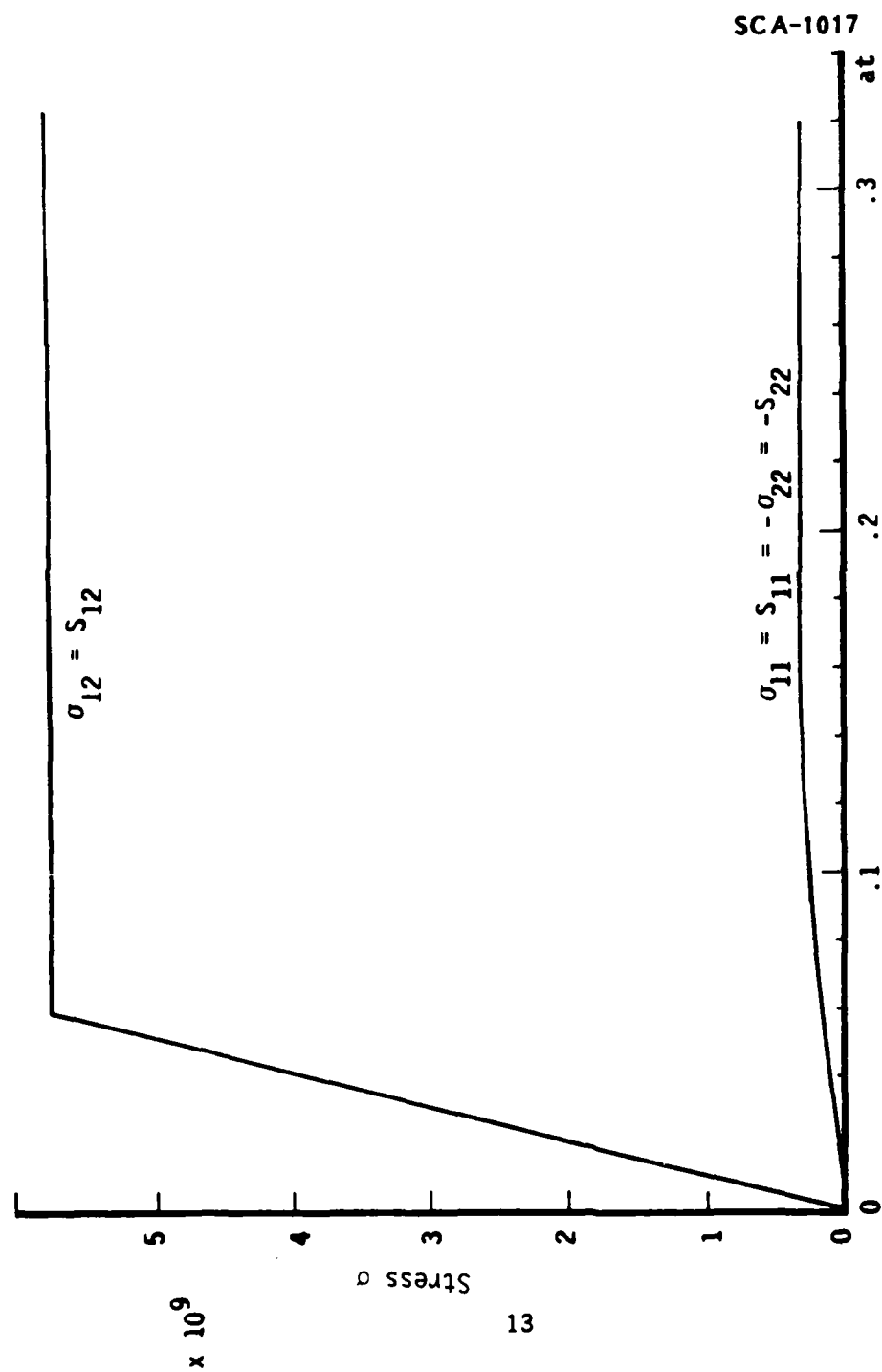


Figure 4. Elastic/plastic solution for $\gamma = \mu/10$: Stress vs. dimensionless time.

SECTION 2
LIST OF REFERENCES

1. Malvern, L.E., Introduction to the Mechanics of a Continuous Medium, Englewood Cliffs, N.J., 1969, Sec. 6.7.
2. Read, H.E., "Various Definitions of Stress Rate, and Their Effect Upon the Constitutive Relation of a Hypoelastic Body of Grade Zero Undergoing Simple Extension," Shock Hydrodynamics, 1965.
3. Prager, W., "An Elementary Discussion of Definitions of Stress Rate," Quarterly of Applied Math. 18, 403-407 (1961).
4. Dienes, J.K., "On the Analysis of Rotation and Stress Rate in Deforming Bodies," Acta Mechanica 32, 217-232 (1979).

DISTRIBUTION LIST

DEPARTMENT OF DEFENSE

ASST TO THE SECY OF DEFENSE ATOMIC ENERGY
ATTN: EXECUTIVE ASSISTANT

DEFENSE INTELLIGENCE AGENCY
ATTN: RTS-2B

DEFENSE NUCLEAR AGENCY
2 CYS ATTN: SPSS
ATTN: SPTD
4 CYS ATTN: STTI-CA

DEFENSE TECHNICAL INFORMATION CENTER
12 CYS ATTN: DD

FIELD COMMAND DNA DET 2
LAWRENCE LIVERMORE NATIONAL LAB
ATTN: FC-1

FIELD COMMAND DEFENSE NUCLEAR AGENCY
ATTN: FCT
ATTN: FCTT
ATTN: FCTT W SUMMA
ATTN: FCTXE

UNDER SECY OF DEF FOR RSCH & ENGRG
ATTN: STRAT & SPACE SYS (OS)

DEPARTMENT OF THE ARMY

HARRY DIAMOND LABORATORIES
ATTN: SCHLD-NW-P

U S ARMY BALLISTIC RESEARCH LAB
ATTN: SLCBR-SS-T TECH LIB

U S ARMY COLD REGION RES ENGR LAB
ATTN: CRREL-EM

U S ARMY ENGR WATERWAYS EXPER STATION
ATTN: D DAY WESSE
ATTN: G P BONNER WESJV-Z
ATTN: J INGRAM WESSER
ATTN: TECHNICAL LIBRARY

U S ARMY MATERIAL COMMAND
ATTN: DRXAM-TL TECH LIB

U S ARMY NUCLEAR & CHEMICAL AGENCY
ATTN: LIBRARY

US ARMY WHITE SANDS MISSILE RANGE
ATTN: STEWS-TE-N K CUMMINGS

DEPARTMENT OF THE NAVY

DAVID TAYLOR NAVAL SHIP R & D CTR
ATTN: CODE 1770
ATTN: TECH INFO CTR CODE 522.1

DEPARTMENT OF THE AIR FORCE

AIR FORCE IS/INT
ATTN: INT

AIR FORCE INSTITUTE OF TECHNOLOGY
ATTN: LIBRARY/AFIT/LDEE

AIR FORCE WEAPONS LABORATORY, AFSC
ATTN: NTE M PLAMONDON
ATTN: NTE J RENICK
ATTN: SUL

AIR UNIVERSITY LIBRARY
ATTN: AUL-LSE

BALLISTIC MISSILE OFFICE/DAA
2 CYS ATTN: ENSN

DEPARTMENT OF ENERGY

UNIVERSITY OF CALIFORNIA
LAWRENCE LIVERMORE NATIONAL LAB
ATTN: L-53 TECH INFO DEPT LIB

SANDIA NATIONAL LABORATORIES
ATTN: EDUC AND TECH LIB DIV

SANDIA NATIONAL LABORATORIES
ATTN: DIV 7111 B VORTMAN
ATTN: TECH LIB 3141 RPTS RECVG CLRK

OTHER GOVERNMENT

CENTRAL INTELLIGENCE AGENCY
ATTN: OSWR/NED

DEPARTMENT OF THE INTERIOR
ATTN: D RODDY

FEDERAL EMERGENCY MANAGEMENT AGENCY
ATTN: OFC OF RSCH/NP H TOVEY

DEPARTMENT OF DEFENSE CONTRACTORS

ACUREX CORP
ATTN: J SAPERSTEIN

DEPT OF DEFENSE CONTRACTORS (CONTINUED)

AEROSPACE CORP
ATTN: LIBRARY ACQUISITION M1/199

APPLIED RESEARCH ASSOCIATES, INC
ATTN: D PIEPENBURG

APPLIED RESEARCH ASSOCIATES, INC
ATTN: R FRANK

BDM CORP
ATTN: A VITELLO
ATTN: CORPORATE LIB

CALIFORNIA RESEARCH & TECHNOLOGY, INC
ATTN: K KREYENHAGEN

CALIFORNIA RESEARCH & TECHNOLOGY, INC
ATTN: F SAUER

CUSHING ASSOCIATES, INC
ATTN: V CUSHING

DEVELCO, INC
ATTN: L RORDEN

GENERAL RESEARCH CORP
ATTN: E STEELE
ATTN: R PARISSE

GEO CENTERS, INC
ATTN: H LINNERUD
ATTN: L ISAACSON

H-TECH LABS, INC
ATTN: B HARTENBAUM

IIT RESEARCH INSTITUTE
ATTN: DOCUMENTS LIBRARY

KAMAN SCIENCES CORP
ATTN: LIBRARY

KAMAN TEMPO
ATTN: DASIAC

KAMAN TEMPO
ATTN: DASIAC

MERRITT CASES, INC
ATTN: J MERRITT
ATTN: LIBRARY

MITRE CORP
ATTN: J FREEDMAN

PACIFIC-SIERRA RESEARCH CORP
ATTN: H BRODE, CHAIRMAN SAGE

PHYSICS APPLICATIONS, INC
ATTN: DOCUMENT CONTROL

R & D ASSOCIATES
ATTN: J LEWIS
ATTN: TECH INFO CENTER

R & D ASSOCIATES
ATTN: G GANONG

RAND CORP
ATTN: P DAVIS

RAND CORP
ATTN: B BENNETT

S-CUBED
ATTN: D GRINE
2 CYS ATTN: D H BROWNELL JR
ATTN: LIBRARY

SCIENCE & ENGRG ASSOC, INC
ATTN: B CHAMBERS

SCIENCE & ENGRG ASSOCIATES, INC
ATTN: J STOCKTON

SCIENCE APPLICATIONS INTL CORP
ATTN: K SITES

SCIENCE APPLICATIONS INTL CORP
ATTN: TECHNICAL LIBRARY

SCIENCE APPLICATIONS INTL CORP
ATTN: W LAYSON

SOUTHWEST RESEARCH INSTITUTE
ATTN: A WENZEL

SRI INTERNATIONAL
ATTN: D KEOUGH
ATTN: P DE CARLI

TELEDYNE BROWN ENGINEERING
ATTN: D ORMOND
ATTN: F LEOPARD

TERRA TEK, INC
ATTN: S GREEN

TRW ELECTRONICS & DEFENSE SECTOR
2 CYS ATTN: N LIPNER
ATTN: TECH INFO CTR DOC ACQ

TRW ELECTRONICS & DEFENSE SECTOR
ATTN: E WONG
ATTN: P DAI

DEPT OF DEFENSE CONTRACTORS (CONTINUED)

WASHINGTON STATE UNIVERSITY
2 CYS ATTN: PHYSICS DEPT Y M GUPTA

WEIDLINGER ASSOC, CONSULTING ENGRG
ATTN: T DEEVY

WEIDLINGER ASSOC, CONSULTING ENGRG
ATTN: M BARON

WEIDLINGER ASSOC, CONSULTING ENGRG
ATTN: J ISENBERG

END

11-86

DTIC